

An Improvement to the Fourier Series Method for Inversion of Laplace Transforms Applied to Elastic and Viscoelastic Waves

by Richard R. Laverty and George A. Gazonas

ARL-RP-160 January 2007

A reprint from the International Journal of Computational Methods, vol. 3, no. 1, pp. 57-69, 2006.

Approved for public release; distribution is unlimited.

NOTICES

Disclaimers

The findings in this report are not to be construed as an official Department of the Army position unless so designated by other authorized documents.

Citation of manufacturer's or trade names does not constitute an official endorsement or approval of the use thereof.

Destroy this report when it is no longer needed. Do not return it to the originator.

Army Research Laboratory

Aberdeen Proving Ground, MD 21005-5069

ARL-RP-160 January 2007

An Improvement to the Fourier Series Method for Inversion of Laplace Transforms Applied to Elastic and Viscoelastic Waves

Richard R. Laverty West Point, NY

George A. Gazonas Weapons and Materials Research Directorate, ARL

A reprint from the International Journal of Computational Methods, vol. 3, no. 1, pp. 57-69, 2006.

Approved for public release; distribution is unlimited.

REPORT DOCUMENTATION PAGE

Form Approved OMB No. 0704-0188

Public reporting burden for this collection of information is estimated to average 1 hour per response, including the time for reviewing instructions, searching existing data sources, gathering and maintaining the data needed, and completing and reviewing the collection information. Send comments regarding this burden estimate or any other aspect of this collection of information, including suggestions for reducing the burden, to Department of Defense, Washington Headquarters Services, Directorate for Information Operations and Reports (0704-0188), 1215 Jefferson Davis Highway, Suite 1204, Arlington, VA 22202-4302. Respondents should be aware that notwithstanding any other provision of law, no person shall be subject to any penalty for failing to comply with a collection of information if it does not display a currently valid OMB control number.

PLEASE DO NOT RETURN YOUR FORM TO THE ABOVE ADDRESS.

1. REPORT DATE (DD-MM-YYYY)	2. REPORT TYPE	3. DATES COVERED (From - To)
January 2007	Reprint	September 2003–April 2005
4. TITLE AND SUBTITLE	5a. CONTRACT NUMBER	
An Improvement to the Fourier S		
Applied to Elastic and Viscoelast	5b. GRANT NUMBER	
		5c. PROGRAM ELEMENT NUMBER
6. AUTHOR(S)	5d. PROJECT NUMBER	
Richard R. Laverty* and George	62105AH84	
	5e. TASK NUMBER	
	5f. WORK UNIT NUMBER	
7. PERFORMING ORGANIZATION NAME	8. PERFORMING ORGANIZATION REPORT NUMBER	
U.S. Army Research Laboratory	ADI DD 160	
ATTN: AMSRD-ARL-WM-MD	ARL-RP-160	
Aberdeen Proving Ground, MD		
9. SPONSORING/MONITORING AGENCY	10. SPONSOR/MONITOR'S ACRONYM(S)	
	11. SPONSOR/MONITOR'S REPORT NUMBER(S)	

12. DISTRIBUTION/AVAILABILITY STATEMENT

Approved for public release; distribution is unlimited.

13. SUPPLEMENTARY NOTES

*Department of Mathematical Sciences, U.S. Military Academy, West Point, NY 10996

A reprint from the International Journal of Computational Methods, vol. 3, no. 1, pp. 57–69, 2006.

14. ABSTRACT

A parametric study of composite strips leads to systems of partial differential equations, coupled through interface conditions, that are naturally solved in Laplace transform space. Because of the complexity of the solutions in transform space and the potential variations due to geometry and materials, a systematic approach to inversion is necessarily numerical. The Dubner-Abate-Crump (DAC) algorithm is the standard in such problems and is implemented. The presence of discontinuous wavefronts in the problems considered leads to Gibbs phenomenon; which, in turn, overestimates the values of maximum stress. These errors are mitigated by use of Lanczos' σ -factors, which combine naturally with the DAC algorithm.

15. SUBJECT TERMS

inverse laplace transform, Gibbs phenomenon, viscoelasticity, waves

16. SECURITY CLASSIFICATION OF:		17. LIMITATION OF ABSTRACT	18. NUMBER OF PAGES	19a. NAME OF RESPONSIBLE PERSON George A. Gazonas	
a. REPORT	b. ABSTRACT	c. THIS PAGE	7.77	2.4	19b. TELEPHONE NUMBER (Include area code)
UNCLASSIFIED	UNCLASSIFIED	UNCLASSIFIED	UL	24	410-306-0863



AN IMPROVEMENT TO THE FOURIER SERIES METHOD FOR INVERSION OF LAPLACE TRANSFORMS APPLIED TO ELASTIC AND VISCOELASTIC WAVES

RICHARD R. LAVERTY

Department of Mathematical Sciences, United States Military Academy
West Point, New York 10996, USA
Richard.Laverty@usma.edu

GEORGE A. GAZONAS

Weapons and Materials Research Directorate, U.S. Army Research Laboratory
Aberdeen Proving Ground, Maryland 21005, USA
gazonas@arl.army.mil

Received 15 April 2005 Revised 15 August 2005 Accepted 15 November 2005

A parametric study of composite strips leads to systems of partial differential equations, coupled through interface conditions, that are naturally solved in Laplace transform space. Because of the complexity of the solutions in transform space and the potential variations due to geometry and materials, a systematic approach to inversion is necessarily numerical. The Dubner-Abate-Crump (DAC) algorithm is the standard in such problems and is implemented. The presence of discontinuous wavefronts in the problems considered leads to Gibbs phenomenon; which, in turn, overestimates the values of maximum stress. These errors are mitigated by use of Lanczos' σ -factors, which combine naturally with the DAC algorithm.

Keywords: Inverse laplace transform; Gibbs phenomenon; viscoelasticity; waves.

1. Introduction

The Laplace transform has proven to be the most natural method for solving the classical initial value problems of dynamic viscoelasticity. The transformation of a viscoelastic initial-boundary value problem (IBVP), by the correspondence principle, is an elastic boundary value problem (BVP), for which the solution is easily constructed. The difficulty then lies in the inversion of this transform. Although many methods exist for numerical inversion of Laplace transforms [Laverty (2003)], the Dubner-Abate-Crump (DAC) algorithm [Crump (1976); Dubner and Abate (1968); Durbin (1974)] has proven to be one of the simplest, yet most robust methods. Its implementation can be achieved in a computer program consisting of just a handful of lines. Its effectiveness can be measured

by the frequency of its use, [Abate and Whitt (1995); Chen and Chou (1998); Frolov and Kitaev (1998); Georgiadis (1993); Georgiadis and Rigatios (1996); Georgiadis et al. (1999)].

Despite all its strengths, the DAC algorithm has one shortcoming within the context of wave propagation, Gibbs phenomenon, [Georgiadis et al. (1999); Laverty (2003)]. The algorithm itself is a construction of an approximate Fourier Series based upon Laplace transform data. Therefore, Gibbs phenomenon will be present at any discontinuity. In general, this will lead to over-estimation of the magnitude of a wavefront in the neighborhood of 10%. This is an unacceptable amount of error when considering optimization problems, such as those considered in [Velo and Gazonas (2003)], where the maximum stress of a two-layered elastic strip is optimized as a function of the impedance ratio of the two materials.

Our goal is to use the most natural analytic construction — the Laplace transform — to examine problems similar to Velo and Gazonas, but using viscoelastic strips. Since analytic inversion of the transforms is impractical for such a large class of problems, we will use the DAC algorithm. The problem of Gibbs phenomenon appears immediately when we attempt to verify the results for the elastic strips. In a review article by Gottlieb and Shu [1997], several methods are described for the mitigation of the Gibbs phenomenon, which were sorted into two classes; filter methods and expansion in orthogonal polynomials. We have found that the filter methods can be implemented very naturally with the DAC algorithm. Furthermore, the use of an expansion in a different basis has two drawbacks from our perspective. First, it requires the computation of the expansion coefficients in the new basis based on the Fourier expansion. Evaluation of these integrals must be done numerically and is a relatively high cost computation compared with the simplicity of the DAC algorithm. Second, the location of discontinuities needs to be known in advance to achieve rapid convergence of the new expansion. We are interested in scattering problems where tracking wavefronts (discontinuities) is not practical. Therefore, we have chosen to mitigate the Gibbs phenomenon via the filter methods; specifically, Lanczos' σ -factors, [Lanczos (1966); Gottlieb and Shu (1997). The adjustment to the standard DAC algorithm is easy to program and does not add appreciably to the computational burden of the inversion.

2. The Elastic/Elastic Strip

The following question was posed (and answered) by Velo and Gazonas [2003]: Can we find an impedance ratio for two perfectly bonded elastic strips such that the maximum stress propagated in each layer (strip) will be a minimum? The answer is yes. In fact, there exists an infinite sequence of discrete impedance ratios that satisfy this requirement. In this section, we attempt to verify this result via the DAC algorithm.

Consider the following coupled initial boundary value problems (IBVPs). The region 0 < x < L/2 will be referred to as layer 1. A step in stress is applied to this layer at x = 0. The displacement in layer 1 is denoted by $u_1(x,t)$ and the elastic

modulus and density are E and ρ , respectively. The partial differential equation (PDE) satisfied by u(x,t) is

$$\rho \frac{\partial^2 u_1}{\partial t^2} = E \frac{\partial^2 u_1}{\partial x^2}, \qquad 0 < x < L/2, \ t > 0, \tag{1}$$

$$0 = u_1(x,0), 0 < x < L/2, (2)$$

$$0 = \frac{\partial u_1}{\partial t}(x, 0), \qquad 0 < x < L/2, \tag{3}$$

$$\Sigma_0 H(t) = -E \frac{\partial u_1}{\partial x}(0, t), \quad t > 0, \tag{4}$$

Layer 2 is the region L/2 < x < L. The displacement in layer 2 is denoted by $u_2(x,t)$, the elastic modulus and density are E/α and ρ/α , respectively. The PDE for $u_2(x,t)$ with the right end (x=L) fixed is

$$\frac{\rho}{\alpha} \frac{\partial^2 u_2}{\partial t^2} = \frac{E}{\alpha} \frac{\partial^2 u_1}{\partial x^2}, \qquad L/2 < x < L, \quad t > 0,$$

$$0 = u_2(x, 0), \qquad L/2 < x < L,$$

$$(5)$$

$$0 = u_2(x,0), L/2 < x < L, (6)$$

$$0 = \frac{\partial u_2}{\partial t}(x,0), \quad L/2 < x < L, \tag{7}$$

$$0 = u_2(L, t), t > 0. (8)$$

To completely determine the solutions to Eqs. (1) through (8) we assume an ideal bonding at the interface: the displacement and stress are assumed continuous at x = L/2.

$$u_1(L/2-,t) = u_2(L/2+,t),$$
 (9)

$$E\frac{\partial u_1}{\partial x}(L/2-,t) = \frac{E}{\alpha}\frac{\partial u_2}{\partial x}(L/2+,t). \tag{10}$$

With these materials the impedance ratio between layers is α and the wave speed, c, is the same in both layers.

$$c = \sqrt{E/\rho}. (11)$$

We investigate how the impedance ratio will effect the maximum stress that will be propagated in each layer. To construct a solution we consider the associated BVPs and interface conditions in Laplace transform space

$$s^2 \hat{u}_1(x;s) = c^2 \hat{u}_1''(x;s), \quad 0 < x < L/2, \quad \frac{\Sigma_0}{s} = -E\hat{u}_1'(0;s),$$
 (12)

$$s^{2}\hat{u}_{2}(x;s) = c^{2}\hat{u}_{2}''(x;s), \quad L/2 < x < L, \ 0 = \hat{u}_{2}(L;s), \tag{13}$$

$$\hat{u}_1(L/2-;s) = \hat{u}_2(L/2+;s), \tag{14}$$

$$E\hat{u}_{1}'(L/2-;s) = \frac{E}{\alpha}\hat{u}_{2}'(L/2+;s), \tag{15}$$

where s is the transform variable, a prime denotes differentiation with respect to x and all transformed quantities are denoted by hats. The solution to Eqs. (12) through (15) can be constructed by elementary means, then the transform of the stress in each layer is given by Eqs. (16) and (17). The stresses will be inverted using the DAC algorithm.

$$E\hat{u}_1'(x;s) = \frac{\Sigma_0}{s} \left(\frac{1-\alpha}{2} \right) \left(\frac{\sinh(\frac{sL}{c})\sinh(\frac{sx}{c})}{\cosh^2(\frac{sL}{2c}) + \alpha\sinh^2(\frac{sL}{2c})} \right) - \frac{\Sigma_0}{s} \cosh\left(\frac{sx}{c}\right), \quad (16)$$

$$\frac{E}{\alpha}\hat{u}_2'(x;s) = \frac{\Sigma_0}{s} \left(\frac{\cosh(\frac{s}{c}(x-L))}{\cosh^2(\frac{sL}{2c}) + \alpha \sinh^2(\frac{sL}{2c})} \right). \tag{17}$$

Figure 1 shows the stress-time history for $\alpha=2$ at the midpoint of each layer. The horizontal line inserted at a stress of $2\Sigma_0$ (relative stress of 2) is of special importance. It has been shown [Velo and Gazonas (2003)] that the stress in layer 1 will never exceed $2\Sigma_0$ and the maximum stress in layer 2 is bounded below by $2\Sigma_0$. This is verified in layer 2, where we see that the maximum stress is clearly greater than this value. However, there are times when the stress in layer 1 is beyond this limit. This is due to Gibbs phenomenon at the discontinuities in stress and is the drawback to the DAC algorithm that we wish to address in this paper.

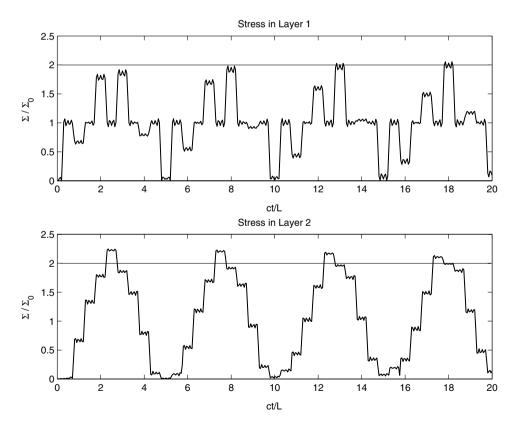


Fig. 1. The time history of stress at the layer midpoints for an impedance ratio of $\alpha=2$ using the DAC algorithm with 256 terms and $tol=10^{-3}$.

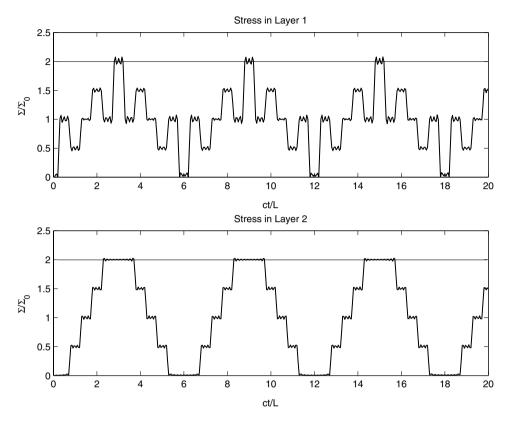


Fig. 2. The time history of stress at the layer midpoints for an impedance ratio of $\alpha = 3$ using the DAC algorithm with 256 terms and $tol = 10^{-3}$.

In Fig. 2 the impedance ratio has changed to $\alpha=3$ and the Gibbs phenomenon in the layer 1 stress is more pronounced. Clearly, the DAC approximation is not faithful to the known bounds of maximum stress in the presence of discontinuous wavefronts.

Figure 3 is a graph of the maximum stress in each layer, as a function of α . The limit of $2\Sigma_0$ is clearly marked and we can see that as α grows there is no simple expression that can capture the values of the maximum stress. There is, however, clear values at which the maximum stress in layer 2 is at a minimum. It can be shown [Velo and Gazonas (2003)] that there exist an infinite number of discrete values of α for which the maximum stress in layer 2 will equal the limiting value of $2\Sigma_0$. Our values never reach all the way down to $2\Sigma_0$ (and our layer 1 values are seldom below $2\Sigma_0$) because of Gibbs phenomenon. Qualitatively, the shape of our graph agrees with that in Velo and Gazonas [2003]; and quantitatively, the critical values of α agree, but the DAC algorithm has not quantitatively captured the optimum values of maximum stress.

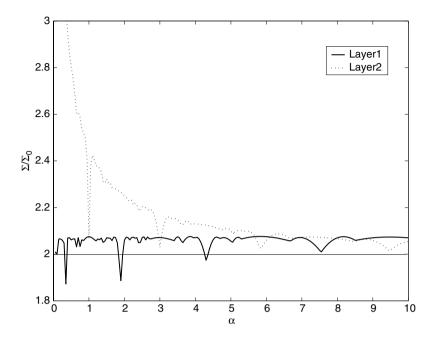


Fig. 3. The maximum stress at the layer midpoints as impedance ratio varies. Solutions were constructed using the DAC algorithm with 256 terms and $tol = 10^{-3}$.

3. Improvement of the DAC Algorithm

Given the Laplace transform $\hat{f}(s)$ we approximate the time domain function f(t) with $\bar{f}(t)$.

$$\bar{f}(t) = \frac{e^{kt}}{T} \left[\frac{\hat{f}(a)}{2} + \sum_{n=1}^{\infty} \text{Re} \left[\hat{f} \left(a + i \frac{n\pi}{T} \right) \right] \cos \left(\frac{n\pi}{T} t \right) - \sum_{n=1}^{\infty} \text{Im} \left[\hat{f} \left(a + i \frac{n\pi}{T} \right) \right] \sin \left(\frac{n\pi}{T} t \right) \right].$$
(18)

When we truncate the series Eq. (18), the function $\bar{f}(t)$ is the DAC approximation to f(t).

Equation (18) is a Fourier Series for the function $\bar{f}(t)$ on the interval (0,T). There are two parameters that we can control to achieve a desired accuracy; the truncation point N and the real number k. Obviously, as we increase N, we will increase the accuracy of our approximation. To achieve a given relative error we choose k according to

$$k = \xi - \frac{1}{2T} \ln(tol), \tag{19}$$

where tol is the bound on the relative error and ξ is a real number chosen slightly larger than the real part of the poles of $\hat{f}(s)$. When we know that f(t) is bounded, we can choose $\xi = 0$.

Equation (18) can be derived directly from a trapezoid rule approximation of the exact inversion integral. However, determination of the parameter k can only be achieved through a more careful analysis based on the periodic extensions of f(t) and the associated exact Fourier Series [Crump (1976); Dubner and Abate (1968); Durbin (1974)].

The Gibbs phenomenon, visible in Figs. 1 and 2, is the source of the errors in Fig. 3 that keep us from making accurate predictions of the smallest maximum stress propagated in layer 2 for a given value of α . However, methods do exist for mitigating the overshoot of Gibbs phenomenon, [Gottlieb and Shu (1997)]. For our purposes, the most effective approach is to implement a filter method. Although many filters exist and are all equally simple to include in our inversion algorithm we have chosen Lanzcos' σ -factors. Its performance makes it a good representative for the general class of filter methods.

To implement the σ -factors, we multiply each coefficient of the Fourier approximation Eq. (18) by a weight σ_n .

$$\sigma_n = \frac{\sin\left(\frac{n\pi}{N}\right)}{\frac{n\pi}{N}},\tag{20}$$

where n is the series index and N is the index value at which we truncate the series. These σ -factors do not effect the convergence of the series, but they do smooth out the Gibbs phenomenon. Figure 4 gives the numerical inversion of Eqs. (16) and (17), using the same numerical parameters and impedance ratio as used in Fig. 2, but including the σ -factors.

In Figs. 2 and 4, we used one of the optimum values of the impedance ratio, $\alpha = 3$. We know that the stress should not cross $2\Sigma_0$ in either layer. It is clear that this fact is verified with our numerical inversion when we use the σ -factors (Fig. 4).

Now, we re-evaluate the maximum stress in each layer as α is varied. Figure 5 is a qualitative and quantitative, faithful reproduction of the analytical results obtained using the method-of-characteristics [Velo and Gazonas (2003)].

4. An Elastic/Viscoelastic Strip

With the confidence that the σ -factors provide us with a means to make accurate inversions, even in the presence of discontinuous wavefronts, we proceed to investigate a composite strip that is composed of an elastic and a viscoelastic material.

Consider a two layered composite occupying the region 0 < x < L where layer 1 is elastic and layer 2 is viscoelastic. We will place their interface at x = l. We maintain the notation that u_1 will be the displacement in layer 1 and u_2 is the displacement in layer 2. However, we must now introduce new notations for the density and stress. The density in layer 1 will be denoted by ρ ; in layer 2 it will be

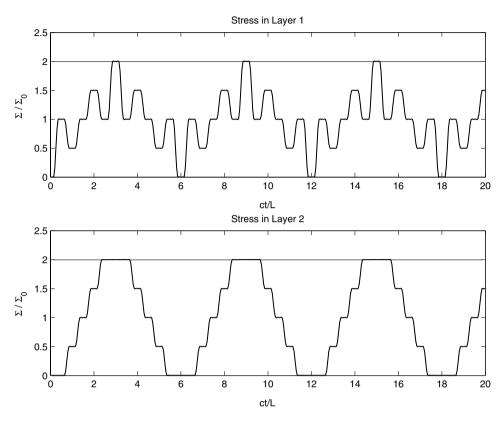


Fig. 4. The time history of stress at the layer midpoints for an impedance ratio of $\alpha = 3$ using the DAC algorithm with Lanczos' σ -factors, 256 terms and $tol = 10^{-3}$.

scaled by a factor of α . The stress functions will be $\Sigma_1(x,t)$ and $\Sigma_2(x,t)$ in layers 1 and 2, respectively. Using this notation, the IBVPs are:

$$\rho \frac{\partial^2 u_1}{\partial t^2} = E \frac{\partial^2 u_1}{\partial x^2}, \qquad 0 < x < l, \ t > 0, \tag{21} \label{eq:2.1}$$

$$0 = u_1(x, 0), 0 < x < l, (22)$$

$$0 = u_1(x, 0), 0 < x < l, (22)$$

$$0 = \frac{\partial u_1}{\partial t}(x, 0), 0 < x < l, (23)$$

$$\Sigma_0 H(t) = -E \frac{\partial u_1}{\partial x}(0, t), \quad t > 0.$$
(24)

$$\frac{\rho}{\alpha} \frac{\partial^2 u_2}{\partial t^2} = \frac{\partial \Sigma_2}{\partial x}, \qquad l < x < L, \ t > 0, \tag{25}$$

$$0 = u_2(x, 0), l < x < L, (26)$$

$$0 = u_2(x,0), l < x < L, (26)$$

$$0 = \frac{\partial u_2}{\partial t}(x,0), l < x < L, (27)$$

$$0 = u_2(L, t), t > 0. (28)$$

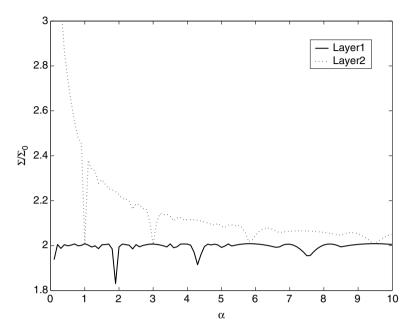


Fig. 5. The maximum stress of each layer (measured at the layer midpoint) as a function of impedance ratio, α , using Lanczos' σ factors. All approximations were made using N=256 and $tol=10^{-3}$.

We need an equation relating the stress τ and displacement w in layer 2. For a linear viscoelastic solid this can be accomplished using an hereditary integral

$$\Sigma_2(x,t) = \frac{\partial u_2}{\partial x}(x,t)G(0^+) + \int_{0^+}^t G'(t-\tau)\frac{\partial u_2}{\partial x}(x,\tau)d\tau, \tag{29}$$

where G(t) is the relaxation modulus and prime denotes differentiation with respect to the argument. Equivalent forms of this constitutive law exist; for instance, after an integration by parts

$$\Sigma_2(x,t) = \int_{0+}^t G(t-\tau) \frac{\partial^2 u_2}{\partial x \partial \tau}(x,\tau) d\tau.$$
 (30)

However, the anticipated discontinuities in strain, w_x , make Eq. (29) more appropriate. There also exist differential forms of the constitutive law, but the convolution form of the hereditary integral formulation is convenient when the Laplace transform is applied.

To complete the model we add the interface conditions

$$u_1(l-,t) = u_2(l+,t),$$
 (31)

$$\Sigma_1(l-,t) = \Sigma_2(l+,t). \tag{32}$$

Taking the Laplace transform of Eqs. (29), (32), and (21) through (28), we get the following BVPs and interface conditions for the transformed displacements.

$$s^2 \hat{u}_1(x;s) = c^2 \hat{u}_1''(x;s), \quad 0 < x < l, \quad \frac{\Sigma_0}{s} = -E\hat{u}_1'(0;s),$$
 (33)

$$s^{2}\hat{u}_{2}(x;s) = \hat{g}^{2}(s)\hat{u}_{2}''(x;s), \quad l < x < L, \quad 0 = \hat{u}_{2}(L;s), \tag{34}$$

$$\hat{u}_1(l-;s) = \hat{u}_2(l+;s), \tag{35}$$

$$E\hat{u}'(l-;s) = \frac{\rho \hat{g}^2(s)}{\alpha} \hat{u}_2'(l+;s), \tag{36}$$

where c is the elastic wave speed, same as (11), $\hat{G}(s)$ is the Laplace transform of the relaxation modulus, and $\hat{g}(s)$ is given by

$$\hat{g}(s) = \sqrt{\frac{\alpha s \hat{G}(s)}{\rho}}. (37)$$

The transformed stresses are

$$\hat{\Sigma}_1(x;s) = -\frac{\Sigma_0}{s} \left[\left(\frac{s^2 \hat{G}(s)}{\hat{g}(s)} \right) \frac{\cosh\left(\frac{s}{\hat{g}(s)}(l-x)\right) \sinh\left(\frac{sx}{c}\right)}{\sinh\left(\frac{sl}{c}\right) d(l)} + \frac{\sinh\left(\frac{s}{c}(l-x)\right)}{\sinh\left(\frac{sl}{c}\right)} \right], \quad (38)$$

$$\hat{\Sigma}_2(x;s) = -\frac{\Sigma_0}{s} \left[\left(\frac{s^2 \hat{G}(s)}{\hat{g}(s)} \right) \frac{\cosh\left(\frac{s}{\hat{g}(s)}(L-x)\right)}{d(l)} \right], \tag{39}$$

where

$$d(l) = \frac{s^2 \hat{G}(s)}{\hat{g}(s)} \cosh\left(\frac{sl}{c}\right) \cosh\left(\frac{s}{\hat{g}(s)}(L-l)\right) + \frac{sE}{c} \sinh\left(\frac{sl}{c}\right) \sinh\left(\frac{s}{\hat{g}(s)}(L-l)\right). \tag{40}$$

Complicated expressions, such as Eqs. (38) through (40), were our original motivation to investigate numerical inversion techniques and have ultimately led us to the DAC algorithm. When we consider the daunting task of analytic inversion of these expressions, and then consider the variations in materials, characterized by G(t), and configurations, more layers and varying widths, pragmatism demands numerical solution. Modification of the most appropriate technique, the DAC algorithm, is the purpose of this current study.

Figure 6 is the result of our modified DAC algorithm applied to Eqs. (38) through (40) using the relaxation modulus

$$G(t) = G_{\infty} + (G_0 - G_{\infty}) e^{-\beta t}.$$
 (41)

Figure 6 also includes the solution for the same problem using the explicit finite element code DYNA3D [Whirley and Engelmann (1993)]. The parameter values are set to: l = L/2, $G_0 = E$, $G_{\infty} = 0.7E$ and $\beta = 1$. We can see in Fig. 6 that the two solution methods independently constructed identical solutions. We can also

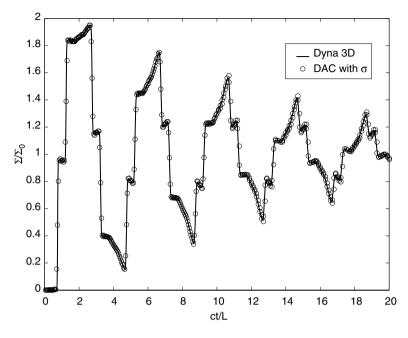


Fig. 6. Corroboration of the stress time history in the viscoelastic layer (measured at the layer midpoint) of an elastic/viscoelastic composite using the DAC algorithm (with σ smoothing) and DYNA3D. Layer 1 is elastic and layer 2 is a standard linear viscoelastic solid. The DAC approximations were made using N=256 and $tol=10^{-3}$.

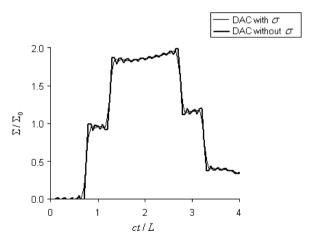


Fig. 7. Stress measured at the midpoint of the viscoelastic layer of the elastic/viscoelastic composite using the DAC algorithm with and without Lanczos' σ -factors.

see the effect of the viscoelastic solid is to "soften" the wavefronts and to dissipate the energy.

Figure 7 focuses on the first peak in Fig. 6. However, in Fig. 7 we have solved the same problem using only the DAC algorithm, with and without Lanczos' σ -factors. It is clear that the Gibbs phenomenon makes a noticable contribution to the peak stress, invalidating the measurement. Thus, the adapted algorithm we employ in this study is an essential part of any quantitative investigation of optimal designs.

5. Conclusion

Any parametric study of composite designs using viscoelastic materials will naturally lead to expressions such as Eqs. (38) and (39). The exact inversion of these transforms is an impractical task. After employing the most robust numerical method available, the DAC algorithm, we found that the Gibbs phenomenon corrupts our results to the extent that they do not match known, analytic solutions. Our search to mitigate these effects produced the Lanczos σ -factors: a general technique developed for use in Fourier series synthesis of functions and completely compatible with the DAC algorithm. The results of the DAC algorithm coupled with the σ -factors verified previous results and have provided means for further studies in viscoelastic composites.

Acknowledgments

This work was performed while the first author (Rich Laverty) held a National Research Council Research Associateship Award at the Army Research Lab and the United States Military Academy.

References

- Abate, J. & Whitt, W. [1995] "Numerical inversion of Laplace transforms of probability distributions," ORSA J. on Computing, 7, 36–43.
- Chen, B. & Chou, T. [1998] "The propagation of one-dimensional transient elastic waves in woven-fabric composites," Composites, Science and Technology, 58, 1385–1396.
- Crump, K. S. [1976] "Numerical inversion of Laplace transforms using a Fourier series approximation," J. ACM 23(1), 89–96.
- Dubner, H. & Abate, J. [1968] "Numerical inversion of Laplace transforms by relating then to the finite Fourier cosine transform," J. ACM 15(1), 115–123.
- Durbin, F. [1974] "Numerical inversion of Laplace transforms: An efficient improvement to Dubner and Abate's method," *Computer J.*, **17**(4), 371–376.
- Frolov, G. A. & Kitaev, M. Y. [1998] "A problem of numerical inversion of implicitly defined Laplace transforms," *Computers Math. Appl.*, **36**(5), 35–44.
- Georgiadis, H. G. [1993] "Shear and torsional impact of cracked viscoelastic bodies A numerical integral equations/transform approach," IJSS, 30(14), 1891–1906.
- Georgiadis, H. G. & Rigatos, A. P. [1996] "Transient SIF results for a cracked viscoelastic strip under concentrated impact loading An integral-transform/function-theoretic approach," Wave Motion, 24, 41–57.

- Georgiadis, H. G. Vamvatsikos, D. & Vardoulakis, I. [1999] "Numerical implementation of the integral-transform solution to Lamb's point-load problem," *Comp. Mech.*, **24**(2), 90–99.
- Gottlieb, D. & Shu, C. W. [1997] "On the Gibbs phenomenon and its resolution," SIAM Review, 39(4), 644–668.
- Lanczos, C. [1966] A Discourse on Fourier Series (Hafner Publishing Company, New York).
 Laverty, R. [2003] "Laplace transform inversion and viscoelastic wave propagation," Proceedings of the 11th Annual ARL/USMA Technical Symposium.
- Velo, A. & Gazonas, G. [2003] "Optimal design of a two-layered elastic strip subjected to transient loading," *IJSS*, **56**(17), 1–10.
- Whirley, R. G. & Engelmann, B. E. [1993] DYNA3D user's manual, a nonlinear, explicit, three-dimensional finite element code for structural mechanics, Lawrence Livermore National Laboratory Report UCRL-MA-107254 Rev. 1.

NO. OF

COPIES ORGANIZATION

1 DEFENSE TECHNICAL (PDF INFORMATION CTR ONLY) DTIC OCA 8725 JOHN J KINGMAN RD

> STE 0944 FORT BELVOIR VA 22060-6218

- 1 US ARMY RSRCH DEV &
 ENGRG CMD
 SYSTEMS OF SYSTEMS
 INTEGRATION
 AMSRD SS T
 6000 6TH ST STE 100
 FORT BELVOIR VA 22060-5608
- 1 DIRECTOR
 US ARMY RESEARCH LAB
 IMNE ALC IMS
 2800 POWDER MILL RD
 ADELPHI MD 20783-1197
- DIRECTOR
 US ARMY RESEARCH LAB
 AMSRD ARL CI OK TL
 2800 POWDER MILL RD
 ADELPHI MD 20783-1197

ABERDEEN PROVING GROUND

1 DIR USARL AMSRD ARL CI OK TP (BLDG 4600)

- 3 DPTY ASSIST SCT FOR R&T SARD TT
 ASA ACT
 T KILLION
 J PARMENTOLA
 C CHABALOWSKI
 THE PENTAGON RM 3E479
 WASHINGTON DC 20310-1714
- 1 US ARMY MATERIEL CMND ASS DPTY CHIEF OF STAFF RSRCH & DEV AMCRDA T R PRICE 9301 CHAPEK RD FT BELVOIR VA 22060-5527
- 1 HQ US ARMY MATERIEL CMND AMCRD 9301 CHAPEK RD FT BELVOIR VA 22060-5527
- 3 AIR FORCE ARMAMENT LAB AFATL DLJW W COOK D BELK J FOSTER EGLIN AFB FL 32542
- 3 DARPA L CHRISTODOULOU W COBLENZ S WAX 3701 N FAIRFAX DR ARLINGTON VA 22203-1714
- 1 DIRECTOR
 US ARMY ARDEC
 AMSRD AAR AEE W
 E BAKER
 BLDG 3022
 PICATINNY ARSENAL NJ
 07806-5000
- 2 US ARMY TARDEC AMSTRA TR R MS 263 K BISHNOI D TEMPLETON WARREN MI 48397-5000

NO. OF COPIES ORGANIZATION

- 4 COMMANDER
 US ARMY BELVOIR RD&E CTR
 STRBE N WESTLICH
 STRBE NAN
 S G BISHOP
 J WILLIAMS
 FORT BELVOIR VA 22060-5166
- 4 COMMANDER
 US ARMY RESEARCH OFFICE
 B LAMATTINA
 A RAJENDRAN
 J LAVERY
 D STEPP
 PO BOX 12211
 RESEARCH TRIANGLE PARK NC
 27709-2211
- 1 NAVAL RESEARCH LAB
 A E WILLIAMS
 CODE 6684
 4555 OVERLOOK AVE SW
 WASHINGTON DC 20375
- 6 DIRECTOR
 LANL
 P MAUDLIN
 R GRAY
 W R THISSELL
 A ZUREK
 F ADDESSIO
 PO BOX 1663
 LOS ALAMOS NM 87545
- 7 DIRECTOR
 SANDIA NATL LABS
 E S HERTEL JR MS 0819
 W REINHART
 T VOGLER
 R BRANNON MS 0820
 L CHHABILDAS MS 1811
 M FURNISH MS 0821
 M KIPP MS 0820
 PO BOX 5800
 ALBUQUERQUE NM 87185-0307
- 2 DIRECTOR LLNL M J MURPHY A HOLT L290 PO BOX 808 LIVERMORE CA 94550

- 3 CALTECH
 M ORTIZ MS 105 50
 G RAVICHANDRAN
 T J AHRENS MS 252 21
 1201 E CALIFORNIA BLVD
 PASADENA CA 91125
- 2 ARMY HIGH PERFORMANCE COMPUTING RSRCH CTR T HOLMQUIST G JOHNSON 1200 WASHINGTON AVE S MINNEAPOLIS MN 55415
- 3 SOUTHWEST RSRCH INST C ANDERSON J WALKER K DANNEMANN PO DRAWER 28510 SAN ANTONIO TX 78284
- 1 TEXAS A&M UNIV
 DEPT OF MATHEMATICS
 J WALTON
 COLLEGE STATION TX 77843
- 6 UNIV OF NEBRASKA
 DEPT OF ENGRG MECH
 D ALLEN
 F BOBARU
 Y DZENIS
 G GOGOS
 M NEGAHBAN
 J TURNER
 LINCOLN NE 68588
- 2 JOHNS HOPKINS UNIV DEPT OF MECHL ENGRG K T RAMESH LATROBE 122 BALTIMORE MD 21218
- 1 WORCESTER POLYTECHNIC INST MATHEMATICAL SCI K LURIE WORCESTER MA 01609
- 3 UNIV OF UTAH
 DEPT OF MATHEMATICS
 A CHERKAEV
 E CHERKAEV
 T FOLIAS
 SALT LAKE CITY UT 84112

NO. OF COPIES ORGANIZATION

- 1 PENN STATE UNIV DEPT OF ENGNG SCI & MECHANICS F COSTANZO UNIV PARK PA 168023
- 3 UNIV OF DELAWARE
 DEPT OF MECH ENGRG
 T BUCHANAN
 T W CHOU
 M SANTARE
 126 SPENCER LAB
 NEWARK DE 19716
- 1 UNIV OF DELAWARE CTR FOR COMPOSITE MATLS J GILLESPIE NEWARK DE 19716
- 3 SRI INTERNATIONAL D CURRAN D SHOCKEY R KLOPP 333 RAVENSWOOD AVE MENLO PARK CA 94025
- 1 VIRGINIA POLYTECHNIC INST COLLEGE OF ENGRG R BATRA BLACKSBURG VA 24061-0219
- 1 COMPUTATIONAL MECHANICS CONSULTANTS J A ZUKAS P O BOX 11314 BALTIMORE MD 21239-0314
- 1 KAMAN SCIENCES CORP D L JONES 2560 HUNTINGTON AVE STE 200 ALEXANDRIA VA 22303
- INST OF ADVANCED TECHLGY
 UNIV OF TX AUSTIN
 S BLESS
 H FAIR
 D LITTLEFIELD
 C PERSAD
 P SULLIVAN
 S SATAPATHY
 3925 W BRAKER LN
 AUSTIN TX 78759-5316

NO. OF NO. OF **COPIES ORGANIZATION COPIES ORGANIZATION** NATL INST OF STAND & TECHLGY APPLIED RSRCH ASSOCIATES D E GRADY BLDG AND FIRE RSRCH LAB 4300 SAN MATEO BLVD NE J MAIN **STE A220** 100 BUREAU DR MS 8611 **ALBUQUERQUE NM 87110** GAITHERSBURG MD 20899-8611 INTERNATIONAL RSRCH ASSOC INC BUCKNELL UNIV D L ORPHAL DEPT OF MECHL ENGRG 4450 BLACK AVE **C RANDOW** PLEASANTON CA 94566 LEWISBURG PA 17837 AKT MISSION RSRCH CORP MATERIALS SCIENCES CORP M EL RAHEB A CAIAZZO 23052 ALCALDE DR R LAVERTY LAGUNA HILLS CA 92653 181 GIBRALTAR RD HORSHAM PA 19044 WASHINGTON ST UNIV SCHOOL OF MECHL & MATL ENGRG J L DING ABERDEEN PROVING GROUND PULLMAN WA 99164-2920 72. DIR USARL WASHINGTON ST UNIV AMSRD ARL WM INST OF SHOCK PHYSICS **PBAKER** Y M GUPTA S KARNA J ASAY J MCCAULEY PULLMAN WA 99164-2814 J SMITH T WRIGHT ARIZONA STATE UNIV AMSRD ARL WM B MECHL AND AEROSPACE ENGRG M ZOLTOSKI D KRAJCINOVIC AMSRD ARL WM BA TEMPE AZ 85287-6106 P PLOSTINS AMSRD ARL WM BC UNIV OF DAYTON J NEWILL RSRCH INST AMSRD ARL WM BD N S BRAR P CONROY 300 COLLEGE PARK B FORCH MS SPC 1911 R LIEB

DAYTON OH 45469

TEXAS A&M UNIV

T GANGI

SCIENCE

A VELO

DEPT OF GEOPHYSICS

UNIV OF SAN DIEGO

5998 ALCALA PARK

SAN DIEGO CA 92110

COLLEGE STATION TX 77843

DEPT OF MATH AND COMPUTER

R PESCE RODRIGUEZ

AMSRD ARL WM BF

AMSRD ARL WM EG

AMSRD ARL WM MA

AMSRD ARL WM MB
M BERMAN
L BURTON
T BOGETTI
W CHOWDHURY
W DE ROSSET

M VANLANDINGHAM

D WILKERSON

E SCHMIDT AMSRD ARL WM M

S MCKNIGHT

R JENSEN

E WETZEL

B RICE

NO. OF COPIES ORGANIZATION

W DRYSDALE A FRYDMAN

D HOPKINS

L KECSKES

THLI

M MINNICINO

B POWERS

J TZENG

AMSRD ARL WM MC

R BOSSOLI

S CORNELISON

M MAHER

W SPURGEON

AMSRD ARL WM MD

B CHEESEMAN

E CHIN

B DOOLEY

C FOUNTZOULAS

G GAZONAS

J LASALVIA

P PATEL

J SANDS

B SCOTT

C F YEN

AMSRD ARL WM RP

J BORNSTEIN

AMSRD ARL WM SG

T ROSENBERGER

AMSRD ARL WM T

J ALTHOUSE

AMSRD ARL WM TA

S SCHOENFELD

AMSRD ARL WM TB

R SKAGGS

J STARKENBERG

AMSRD ARL WM TC

R COATES

K KIMSEY

D SCHEFFLER

S SCHRAML

AMSRD ARL WM TD

S BILYK

T BJERKE

D CASEM

J CLAYTON

D DANDEKAR

M GREENFIELD

K IYER

B LOVE

M RAFTENBERG

E RAPACKI

M SCHEIDLER

S SEGELETES

T WEERASOORIYA

AMSRD ARL WM TE J POWELL B RINGERS G THOMSON

- 1 DERA
 N J LYNCH
 WEAPONS SYSTEMS
 BLDG A20
 DRA FORT HALSTEAD
 SEVENOAKS
 KENT TN 147BP
 UNITED KINGDOM
- 1 ERNST MACH INTITUT H NAHAME ECKERSTRASSE 4 D 7800 FREIBURG 1 BR 791 4 GERMANY
- 1 FOA2 P LUNDBERG S 14725 TUMBA SWEDEN
- 1 PCS GROUP
 CAVENDISH LABORATORY
 W G PROUD
 MADINGLEY RD
 CAMBRIDGE
 UNITED KINGDOM
- 1 CENTRE D ETUDES DE GRAMAT J Y TRANCHET 46500 GRAMAT FRANCE
- 1 SPART DIRECTION BP 19
 DR E WARINGHAM
 10 PLACE GEORGES
 CLEMENCEOUX
 92211 SAINT CLOUD CEDEX
 FRANCE
- 1 LMT CACHAN
 J F MOLINARI
 61 AVE DU PRESIDENT WILSON
 94235 CACHAN CEDEX
 FRANCE
- 1 TECHNICAL UNIV OF CRETE DEPT OF MINERAL RES ENGNG G EXADAKTYLOS CHANIA CRETE GREECE

NO. OF COPIES ORGANIZATION

- 1 ROYAL MILITARY COLLEGE OF SCIENCE CRANEFIELD UNIV J MILLETT SHRIVENHAM SWINDON SN6 8LA UNITED KINGDOM
- 1 UNIV OF MANCHESTER
 N K BOURNE
 PO BOX 88
 SACKVILLE STREET MANCHESTER
 M60 1QD UK
- 1 BEN GURIAN UNIV OF NEGEV E ZARETSKY DEPT OF MECH ENG BEER SHEVA ISRAEL 84105
- 2 RUSSIAN ACADEMY OF SCIENCES INSTITUTE FOR HIGH ENERGY DENSITIES G I KANEL S V RAZORENOV IVTAN IZHORSKAYA 13/19 MOSCOW 127412 RUSSIA
- INST FUR
 MATERIALFORSCHUNG II
 C TSAKMAKIS
 POSTFACH 3640
 FORSCHUNGSZENTRUM
 KARLSRUHE
 D 76021 KARLSRUHE
 GERMANY
- 2 NATL TECH UNIV OF ATHENS DEPT OF ENG SCI H G GEORGIADIS I VARDOULAKIS ATHENS 15773 GREECE
- 1 UNIV OF PATRAS DEPT OF CIVIL ENGRG D BESKOS PATRAS 26500 GREECE
- ARISTOTLE UNIV
 OF THESSALONIKI
 DEPT OF MECH & MATLS
 E AIFANTIS
 THESSALONIKI 54006
 GREECE

DEMOCRITUS UNIV OF THRACE DEPT OF CIVIL ENGRG E GDOUTOS XANTHI GREECE